

AVIATION

The Oldest American Aeronautical Magazine

JULY 27, 1925

Issued Weekly

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Macready climbing for his altitude record

VOLUME
XIX

SPECIAL FEATURES

NUMBER
4

SUPERCHARGERS

THE R.A.F. DISPLAY

THE AIR MAIL ROUTES

GARDNER PUBLISHING CO., INC.

HIGHLAND, N. Y.

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JULY 27, 1925

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Published every Monday

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Wing span	32' 6"	Wing area	320 sq. ft.
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Wing	80' 15"	Trailing edge	1.50 in.

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JULY 27, 1925

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RALPH H. UPTON, BUSINESS MANAGER
EDWARD T. ALLIS, CIRCULATION MANAGER

No 4

Great Silence

In the early days of the automobile, while the chain drive was still used and proper methods had not been invented, one often heard people say how noisy it would be if such construction came into general use. Automobiles have come, but they have diversified the possibilities of the early passenger by having greater than four driven vehicles. The silencing of the automobile was accomplished deliberately, but much of it came about because noise meant finance and wear and the quiet engine was more efficient.

Today, when a Martin Bomber or an F8L comes overheard, swishing its early morning shadow, at clutching the pause of an evening, even an aviation enthusiast has visions of a terrible future when the whole heaven will pulsate to the whining beat of propellers. At present, aviation has more fundamental problems to consider than the intricacies of mere ground running sounds, but once these problems are on the way to solution, the problem of noise will assume greater importance. In fact, even now, the deafening noise of traveling by plane is an important factor in deterring passengers from commercial air lines. A silent plane would also have great advantages in noiseless gas and gasoline storage tanks, and an airplane would be a great deal more if the ground troops were not aware of this present handicap.

Low statics and swiveling engines have quieted the power plant, but the aerodynamic noise which may be the key to much inefficiency is still prettily an unbroken field. The whirr of a propeller, or its pop and crackle when its peripheral speed approaches that of sound, is undoubtedly an indication of a lack of efficiency.

When the power plant is functioning the whirling of the wires is not noticed, but as a glide can have a variety of sounds which indicate a broken air flow and wasted power, just as certainly as a squeak in an unbalanceable automobile friction and wear. Of course, it does not take much power to produce these sounds but as we reach ultimate refinement in design, all the available power will be utilized and airplanes will become quieter.

New Aircraft

It has always been a feature of the annual Royal Air Force Displays to have a "fly past" of new types of aircraft. In recent years the custom has arisen of making this the first public announcement of the new types that are developed. This is in correspondence with this country. Several new designs have appeared at some of the national air meets but these machines were there as exhibits in some event or merely as transports for some officer, not primarily to be seen.

It would be an excellent thing if such a "fly past" were made a feature of our national air meets. While it is true

that our service squadrons are mounted on revolutionized wartime machines for the most part, it is not true that these let us have no development of new types. Our latest pursuit planes are the best in the world. The winner and the runner-up in the recent observation plane contest are superior to those of any other air power. Several very interesting designs for attack planes, transports, and bombers have been produced. If the best of these machines were given a public showing, the public would have more faith in American designers and would be induced to inquire why the military and naval air services are not better equipped.

Radio

ONE of the features of the aerial participation in the Joint Army and Navy Exercises at Hawaii was the extensive use of radio, both in the air and on the ground. Major Street stated in his review of the problems that the Air Service stationed at Hawaii should devote considerable time to radio training. This is true in this country as well. Air Service personnel seem to regard radio as being all very well in its way but of doubtful use to them.

This attitude is not the fault of the members of the Service Squadrons. As has been the case with practically all other military services, the War and Navy Air Services are enormous sources of war-time radio operations. This operation was the best in the world in 1918, but is quite outside now. In consequence of our policy of using up this material in the interest of economy, and building new designs at experimental quantities only, the average squadrons often are not aware of the tremendous advances that have been made.

Modern equipment is now being used for service use. The new sets that have been tuned to the flying fields are being used in an increasing variety of ways, as the squadrons become acquainted with their capabilities. One of these has been the reporting of events and the transmission of the reports by broadcasting stations. The use of reliable radio apparatus is modifying aerial observation and spotting by permitting much more accurate reports and a more rapid interchange of signals than has been the case heretofore.

Another application of modern equipment is voice controlled cameras. This is itself nothing new. President Wilson gave orders for a filmatic in the air in 1918. This was only in the nature of a demonstration. The equipment then available did not permit of this being done as a regular part of the work over the lines. In fact, it was only in the later stages of the war that wireless firms became compulsory. In the next war, after aircraft production gets started, anything smaller than a squadrons will very likely be set off and destroyed. Radio is a necessity for the efficient control of this number of machines. The air force that has the best developed radio, both material and operating personnel, will have a great advantage.

Superchargers

By DAVID GREGG

Research Expert, Engineering Division, McCord Field

The basic factor which determines the power output of an internal combustion engine is the weight of fuel burned in a given time. The engine's power output is proportional to the weight of fuel burned between the first and the next stroke it may have and that the power depends directly on the weight of air used. The volume of air used at each stroke is constant, but the weight depends upon the engine speed. Therefore, increasing, remains through the intake valves, etc.

Concerning with the development of the intake combustion engine, there have been many attempts to increase the weight of air burned in any given engine by some external device. For example, most Diesel engines have separate cylinders for compressing and delivering air to the power cylinders in excess of the quantity that these cylinders would adequately serve. A supercharger is simply an external air pump designed to supply this extra air. In the aircraft engine, it is used to increase the indicated output of an engine when altitude is reached and the surrounding atmospheric pressure decreases.

For a supercharger to be economically applied to an airplane engine, three factors must be considered, namely weight, cost and fuel consumption. The Liberty engine weighs 1,000 lb. and fuel consumption is 20 lb. per horsepower at 1,000 rpm. At 20,000 ft. the basic horsepower is only 18.83 or 35.7 per cent of the sea level power. The weight per horsepower at this altitude is 56.6 lb. per hp. The actual weight of a supercharger to maintain approximately sea level power on the Liberty engine is 882 lb. which includes the increased weight of the auxiliary system. The weight per horsepower of the uncharged Liberty engine is 1862 lb. for the power output at 20,000 ft. is 696 hp., and the weight per horsepower is 37.0 lb., or less than one-half of that of the uncharged engine. The power of the uncharged Liberty would have to be increased to 248 per cent to have it up to that of the supercharged engine. It is obvious that such an increase would be accompanied without greatly increasing the weight and bulk of the engine.

In fuel consumption the supercharged engine has a decided advantage in the constant density of intake air maintained by the supercharger keeps the specific fuel consumption at approximately the normal rate, while the specific fuel consumption of the uncharged engine increases 22 per cent at 20,000 ft.

From the foregoing, it is apparent that a supercharger can be economically applied to an aircraft engine and is an actual necessity for high altitude flying.

Classification

As a supercharger is nothing more or less than an air pump, there are several types available for consideration. First, is the pump type or centrifugal. The capacities at low speeds are necessarily relatively small, and, although having a high efficiency, a centrifugal is large and heavy to be considered for aircraft purposes.

The second classification is the rotary blower, such as a gear pump, vane pump or Roots blower. Pumps of this type are not so heavy than the piston type, operate at speeds as high as 50,000 rpm. and are relatively inexpensive per unit of power used. The Roots blower has more difficulty in maintaining a constant flow to the power train necessary to give the two rotating impellers the proper angular reaction. The clearances must be sufficiently small to prevent the impellers from rubbing together, or from running against the compressor housing and yet must be small enough to prevent excessive leakage of air. As the pressure rises with the type of pump the dependence of speed, as it is used with most of these, is an important factor.

The third classification is the centrifugal compressor which consists simply of an impeller rotating in a closed casing. It is extremely simple mechanically as it has only one moving

part, it operates at very high speeds and has a slightly better efficiency than the rotary blower. The type of pump used for the compressor can be selected in connection with the engine. In the smaller sizes, however, it is possible to use one separate driving engine such as a 350 hp. Hispano engine or smaller engine with a pump which supplies air both to the driving engine and to the other engine of the airplane. A separate drive of this nature is equally as reliable as the other engine installed in the airplane and the failure of the supercharger will not affect the operation of the engine.

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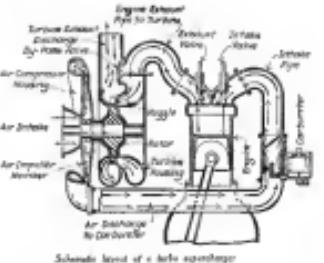
The second classification is the rotary blower, such as a gear pump, vane pump or Roots blower.

fact that they are directly geared to the engine causes frequent fatigue failure due to the transmission of regular vibration and torsional vibration of the crankshaft through the gear train. In the smaller sizes, geared superchargers have operated quite successfully without mechanical difficulties and have shown remarkable reliability over long periods of time.

Another method of drive for the engine is the belt drive developed by M. B. Birkett of England and is shown in the diagrammatic fig. 1 for the Air Service. It offers probably the simplest and most reliable method of drive for an airplane supercharger. The speed of the gas turbine is very high, coinciding with the necessary speed of a centrifugal compressor, and as developed by the Engineering Division at the present time the supercharger can be driven in this manner without difficulty. The drive is simple, reliable and the cost of the engine is not increased by the addition of the supercharger. The cost of the engine itself is the same as the cost of a turbo supercharger but the exterior heat of the exhaust gas. This is in the neighborhood of 1,000° F. Birkett has recommended with the most fields and small losses used to convey the gas in the turbine wheel. The losses between the various parts of the nozzle, wheel, nozzle and exit, the nozzle exit, the nozzle ring, wheel, nozzle and exit, and the turbine wheel, respectively, are at least 20,000 rpm. were never attained and in the design limit of their material was not used. The nozzle exit velocity of the gas when they attained the highest frequency, pitch and yaw, which they recommended, and difficulties were met with the bearings, methods of mounting, and hence construction details.

Difficulties Removed

The Engineering Division after five or six years steady development and experimental work on the turbo supercharger has removed the difficulties which were inherent in the early designs. The difficulties have been removed by the use of a low atmospheric pressure of the exhaust pressure which permits the nozzle exit pressure which properly compensates for the power required to drive the supercharger. At lower altitudes, however, this is not the case. Take for example flying just at sea level. There is a maximum power required to drive the supercharger which is about 3000 ft. above sea level. At sea level the supercharger is in a position to develop added power in the engine, as the actual output of the engine will be 420 hp. less the maximum power required to drive the supercharger. In other words, a given drive supercharger develops less than normal horsepower at low altitudes and will maintain approximately full normal horsepower at the rated altitude of the supercharger.



The superchargers may also be geared directly to the engine as superchargers such as in this case, its reliability is in doubt upon the gas flow driven. A flywheel, however, such as the one used in the blower is necessary to maintain speed when a steady flow of air is on one side of the gas turbine. The Roots blower, for example, is necessary to provide the proper angular rotation of the blower and keep the impellers from rubbing. These gears are necessary regardless of whether or not the blower is driven at approximately the higher speeds of the supercharger. The gear drive is the most reliable and the most efficient. The American Society of Mechanical Engineers Committee on this type of blower, it is not yet possible to draw any definite conclusion as to its ultimate success. In large sizes, and it certainly affords more mechanical complications than the centrifugal type.

When the centrifugal blower is driven by direct gearing from the engine a gear ratio of from 1 to 1.5 to 1 is necessary to maintain the blower at speeds between 20,000 and 30,000 rpm. in order to reduce the size of the compressor and to provide a sufficient compression ratio. The high speed of the impeller when geared directly to the engine introduces considerable difficulties in the gear train and bearings due to the inertia of the moving parts. The Engineering Division has developed a centrifugal compressor which can be driven by a motor having no difficulty in the gear train operating at speeds as high as 7,000 ft./min. Ball and roller bearings have been used in these installations to maintain accurate gear ratios and to reduce the friction and lubricating difficulties. The problem of ball bearings at these high speeds is one of unusual stress in the bearing itself rather than in the load imposed by the weight of the impeller. It is yet to be determined whether or not it will be possible to satisfactorily operate anti-friction bearings at such high speeds and still have them thoroughly bearings over a long period of time. The inertia of the impeller and the moving parts and the



Liberty type supercharger of the General Electric Co., installed on a 2444 plane of the Air Service Engineering Division.

The turbo supercharger derives its power from the residual heat and energy left in the exhaust gas at the end of the expansion stroke. The gas is exhausted in a most efficient, discharged through a series of nozzle jets a turbine wheel mounted on the same shaft as the impeller. At altitude, the exhaust gas

is kept slightly above sea level pressure, and expands through the nozzle to an atmospheric pressure of approximately 7 lb. per sq. in. The energy taken from the gas during this pressure drop is sufficient to drive the turbine and supply a full charge of intake air at sea level pressure. The cost of the engine is of a type which allows the greatest portion of the engine to be used directly to the supercharger. At the present time the rate of intake air is about the same as the exhaust air discharged at atmospheric pressure, so that the supercharger is placed on the engine at sea level. It should also be noted that at altitude the power output of the engine does not exceed its normal output on account of the supercharger, whereas at the same altitude the intake air is not taken back into the engine, so that the engine is not able to use the maximum horsepower required to drive the supercharger.

The first problem in the design of a turbo supercharger was the exterior heat of the exhaust gas. This is in the neighborhood of 1,000° F. Birkett was recommended with the most fields and small losses used to convey the gas in the turbine wheel. The losses between the various parts of the nozzle, wheel, nozzle and exit, the nozzle exit, the nozzle ring, wheel, nozzle and exit, and the turbine wheel, respectively, are at least 20,000 rpm. were never attained and in the design limit of their material was not used. The nozzle exit velocity of the gas when they attained the highest frequency, pitch and yaw, which they recommended, and difficulties were met with the bearings, methods of mounting, and hence construction details.

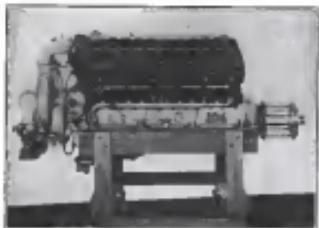
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When air is compressed its temperature is raised due to the loss of available compression. As the supercharger is not 100 per cent efficient, the turbines, friction losses, etc., reduce the temperature of the compressed air. When the air was not cooled, the temperature was raised to such an extent that the insulation would melt on the hot turbine housing at altitude.

The obvious solution was the addition of an air cooler, located between the supercharger and the turbine, to reduce the air temperature to a point at which the engine could run satisfactorily without preheating. In this way the insulation was not melted, the maximum temperature of altitude is below the melting point of the insulation. It has been found that metal electrodes with solid electrodes form better insulation, and are themselves a contributing factor to preheating. Plugs with larger electrodes having adequate cooling surface overruns

gas turbine. In a number of cases where spark plugs with the porosity insulation were used, the porcelain cracked, pieces fell in the engine cylinder and were blown out with the exhaust gases into the turbine wheel and the pieces of porcelain striking the blades caused at very high velocity especially twisted them. For this reason, non-plugs are used on all turbo-supercharged engines.



Curtiss D-2 aircraft engine.

The next serious difficulty was with the fuel system. On the old machines the pressure system was well laid out so the engine would run well and the pressure system was required in the gas tanks to force the fuel in the carburetor against the supercharger pressure. Fuel pumps were then tested but it was found that they would not work satisfactorily unless they were located lower than the level of the fuel in the tank, so at first the fuel tank would operate well and could be located to the rear of the engine. The present arrangement of fuel system consists of a gear pump driven by a flexible shaft and located lower than the fuel tank. Fuel flows to the pump by gravity and is pumped through a supercharger relief valve to the carburetor. This relief valve consists of a spring loaded valve on which is maintained the same air pressure as the air entering the carburetor. The valve is set in the air line from the rear so will be equipped with some type of supercharger.

The additional power output of a supercharged engine requires approximately 20 per cent increase in the radiating surface, so far, although the air intake area of altitude is much larger than the intake area of the engine, the intake air is much less than the engine takes per square foot of radiator surface decreased. Eighty degrees Centigrade is an ideal operating temperature for an altitude engine. If the temperature is maintained at high altitudes, a panel would show the water temperature is high due to the lower atmospheric pressure. It is therefore necessary to have the entire cooling system and have a pressure relief valve which will open with a pressure greater than atmospheric if maintained.

The losses of a supercharged engine is generally constant with increasing altitude and as the propeller torque decreases with altitude, the propeller will speed up. For this reason a propeller must be used which holds down the engine rate of climb of the aircraft and allows it to speed up and maintain its normal rate at the designed altitude of the aircraft.

In the earlier series no suitable oxygen system was available and a small hand carried system was made up using compressed oxygen in bottles. The pressure was reduced as necessary to a Regal valve which reduced the oxygen delivered directly to the pilot's mouth through a flexible tube. This system is reliable and, although somewhat wasteful of oxygen, it has not been replaced up to the present time although the Material Section has selected mass oxygen with a liquid oxygen system, which is more pleasant to use than the compressed oxygen as there is no taste whatever.

In order to reduce the pilot of the necessity of continually watching his supercharger operation, an automatic control has been devised which will maintain air level pressure at any altitude without attention on the pilot's part. The first of these controls worked satisfactorily up to 20,000 ft or 12,000 ft, but was not sufficiently powerful to maintain constant altitude above this altitude. Tests are being carried on a new system to hold constant altitude. The author is in the light of present knowledge which will undoubtedly improve the pilot of the necessity of continually adjusting the supercharger.

In connection with the supercharged two speed instruments are used. One is a differential fuel pressure gauge which shows the difference between the fuel pressure in the feed line and the carburetor float chamber. The other is a pressure gauge which shows the supercharger air pressure either in lb/sq. in. or in. of water. The former gauge is now replacing the latter type and the pilot by operating the supercharger control maintains the carburetor pressure at air level, 14.7 lb. per sq. in.

Results

The application of the supercharged engine to aeronautics increased the performance which it was possible to obtain. For example the first installation was made on an Avro having a 260 hp. Hispano engine. The ordnance of the ship unsupercharged was approximately 22,000 lb., and with the supercharger installed the aircraft had a range of 1,000 miles at 14,000 ft to approximately 28,000 lb. In case of the supercharger, while the ceiling with full military load has been increased from 7,000 ft to better than 13,000 ft. A DH 81 with a military load and supercharger will reach an altitude of 20,000 ft at 14,000 ft. All turbo-supercharged aircraft have been shown to costing airplanes, and that, with one exception, airplanes have been designed to take full advantage of the supercharged engine. It is therefore reasonable to assume that even better results will be obtained when this is done.

You can remember the Klemm Boulton C.9, which was designed especially for the supercharged engine, especially, broke the world's altitude record, curving up to 22,000 ft at 14,000 ft. In addition, it had a ceiling of around 31,000 ft. At the present time all new aeronautic engines are being designed to accommodate either a gear driven or turbine supercharger and practically all engines that are used in the Air Service from now on will be equipped with some type of supercharger.

Model Enthusiasts in England



A. A. Price

Weighting in the machine at the Annual Derby of the British Society of Model Aeronautical Engineers

The New Mail Routes

Details of the Advertisements for Bids for the First Routes to be operated under the Kelly Bill

Postmaster General has issued advertisements for bids for eight new or new routes as July 15. Sealed proposals for these routes will be received at the Post Office Department, Postmaster General, until 4:30 p.m. on Sept. 15, 1935. A bond of \$2,000 is required with each bid. The validity for these routes is embodied in an Act of Congress approved Feb. 2, 1935 and commonly known as the Kelly Bill. The detailed regulations were given on pages 458 and 459, Vol. XXVII of the April 1, 1935 issue of *Aviation*.

The routes are all designed to coincide with the present air mail services operated by the carriers. The schedules have been prepared requiring an average flying speed of approximately 80 m.p.h. The department cautions that in some instances, due to weather conditions, etc., it may be impossible to maintain such average speed on the other hand when conditions are favorable it may be possible to cover the distance in even less time. Payment allowances will be made on all one-way routes. The schedules follow:

DETROIT, MICH., TO BARTON, COLOR. TO NEW YORK, N. Y.
AND RETURN

Route: West by Indianapolis, Ind.; via Toledo, Ohio, and Detroit, Mich., to New York, N. Y.

Leave Detroit 10:30 a.m. via Toledo, Ohio, and New York 4:15 p.m. and will reach New York 6:15 p.m. Return, 4:15 p.m. via Toledo, Ohio, and Detroit, Mich., and New York 6:15 p.m.

DETROIT, MICH., TO INDIANAPOLIS, IND.; LOUISVILLE, KY. AND MARSHFIELD, WIS., TO BIRMINGHAM, ALA. AND RETURN

Route: West by Indianapolis, Ind.; via Toledo, Ohio, and Detroit, Mich., to Louisville, Ky., and New York 4:15 p.m. Return, 4:15 p.m. via Toledo, Ohio, and Detroit, Mich., and New York 6:15 p.m.

DETROIT, MICH., TO CHICAGO, ILL., TO ST. PAUL AND MINNEAPOLIS, MINN., AND RETURN

Route: West by Toledo, Ohio, and Detroit, Mich., to Chicago, Ill., and New York 4:15 p.m. Return, 4:15 p.m. via Chicago, Ill., and Detroit, Mich., and New York 6:15 p.m.

DETROIT, MICH., TO NEW YORK, N. Y.; ST. JOSEPH AND KANSAS CITY, MO., AND RETURN

Route: West by Toledo, Ohio, and Detroit, Mich., to St. Louis, Mo., and New York 4:15 p.m. Return, 4:15 p.m. via St. Louis, Mo., and Detroit, Mich., and New York 6:15 p.m.

DETROIT, MICH., TO SPRINGFIELD, ILL., TO ST. LOUIS, MO., AND RETURN

Route: West by Toledo, Ohio, and Detroit, Mich., to St. Louis, Mo., and New York 4:15 p.m. Return, 4:15 p.m. via St. Louis, Mo., and Detroit, Mich., and New York 6:15 p.m.

DETROIT, MICH., TO LOS VEGAS, NEV., TO LOS ANGELES, CALIF., AND RETURN

Route: West by Toledo, Ohio, and Detroit, Mich., to St. Louis, Mo., and New York 4:15 p.m. Return, 4:15 p.m. via St. Louis, Mo., and Los Angeles, Calif., and New York 6:15 p.m.

DETROIT, MICH., TO PORTLAND, ORE., TO SEATTLE, WASH., AND RETURN

Route: West by Toledo, Ohio, and Detroit, Mich., to St. Louis, Mo., and New York 4:15 p.m. Return, 4:15 p.m. via St. Louis, Mo., and Los Angeles, Calif., and New York 6:15 p.m.

DETROIT, MICH., TO ST. PAUL, MINN., TO LOS ANGELES, CALIF., AND RETURN

Route: West by Toledo, Ohio, and Detroit, Mich., to St. Louis, Mo., and New York 4:15 p.m. Return, 4:15 p.m. via St. Louis, Mo., and Los Angeles, Calif., and New York 6:15 p.m.

DETROIT, MICH., TO PORTLAND, ORE., AND RETURN

Route: West by Toledo, Ohio, and Detroit, Mich., to St. Louis, Mo., and New York 4:15 p.m. Return, 4:15 p.m. via St. Louis, Mo., and Los Angeles, Calif., and New York 6:15 p.m.

DETROIT, MICH., TO LOS ANGELES, CALIF., AND RETURN

Route: West by Toledo, Ohio, and Detroit, Mich., to St. Louis, Mo., and New York 4:15 p.m. Return, 4:15 p.m. via St. Louis, Mo., and Los Angeles, Calif., and New York 6:15 p.m.

The department reserves the right to modify or change these schedules or the route of the mail, discontinued and operating maillines, will postpaid. Proprietary marks may be obtained from postmasters on the various routes, or from the office of the Second Assistant Postmaster General.

On the route as which is located Dallas and Fort Worth, it is intended that a landing field for receiving and delivering mail shall be located approximately mid-way between these cities. This also applies to the use of St. Paul and Minneapolis.

Carriers should state the number of planes it is proposed to provide for carrying the mails on the route (including the number in reserve for each plane in the set), description of planes, showing number of engines, horsepower, speed, cruising altitude, etc.

Carriers and their agents are urged to acquaint themselves fully with the laws of Congress relating to aeronautics for navigating the route as to dimensions and with the requirements set forth in this advertisement and with the service to be performed before they assume any liability as holders or carriers, and to prevent misappropriation of mail of commercial性质.

Carriers will be required to furnish bond with type or more individual carriers approved by a postmaster of the first, second, or third class, which bond must be acceptable to the Postmaster General, such service to be shown to the owners of real estate situated along all conceivable at an amount double the amount of the bond required, or a surety company which has complied with the provisions of the Act of June 25, 1934, and is in good standing. The bond of surety company mentioned may be found on page 46 of the Official Postal Manual for July, 1935, a copy of which is furnished each postmaster.

Mail compartments in all planes to be made disappear to the satisfaction of the Post Office Department. When a bid is signed by an aeronautical company, it should be signed in the event of the death of the person signing the bid to so sign on behalf of the company. Such evidence may be a certified copy of the postmaster general to whom the person was elected, an appraisal as officer or agent of the company and that part of the by-laws showing his authority as such officer or agent to sign the bid, or a certified copy of a resolution of the board of directors or managers of the company.

Bids should be for service as stated in the advertisement.

A proposal entered in any of the essential terms can not be considered in competition with bids submitted in proper form.

The accepted bidder shall execute the formal contract prepared by the Department, naming service advertised, with good and sufficient sureties acceptable to the Postmaster General, and the such contracts within thirty days from date of acceptance of the proposal. The Postmaster General reserves the right to reject all bids.

If the accepted bidder fails to file a properly executed contract, or fails to make the same in the Department within thirty days from the date of acceptance of the bid, or having executed the contract fails to provide the service required, he may be declared a failing bidder or contractor and proceedings taken according to law.

Bids should be forwarded in time to reach the office of the Second Assistant Postmaster General not later than 4:30 p.m. on Sept. 15, 1935. Bids received after this date will be considered in competition with bids made within the prescribed time.

Bids should be sent in sealed envelope, unaddressed "Aeronautic Proprietary," and addressed to the "Second Assistant Postmaster General, Post Office Department, Washington, D. C."

Aviation in Honduras

A Frank Statement of Actual Conditions in the Republic of Honduras in Response to Many Requests

By CHRISTOPHER V. PICKUP

The hundred or so visitors and people referred to in this, who have written me for information during the past few months, have generally begun their questions by asking what types of planes are in use here. I shall try to the best of my ability to answer all the questions honestly and impartially. The first question is, what is the type of plane used by the French Standard Aeroplane Company at the present time? While I have seen one in November last, the Tissandier Company had but one plane, an L-50 which later turned to Teigruppe. The loss was due to a defect in the gas burner connection on the carburetor, and was not the fault of the plane or engine, being merely a weak place on the plane. That type of plane is very well in all its planes but not in Bloch's at the present time. There are, however, two others to previous record, the first being the model of the German flying boats. There are at Teigruppe, the capital, one at St. Pauli, one at Dass and the other at Le Coq. The last two were mentioned as seats on the north coast.

"We are now less two of the Tournaments with those engines. They are in again against the only proved plane for us here. They carry all the load the small fields will permit and cannot be beaten in the use of those engines. But it stops us out of the main line of travel. We have to go a long while to come, due to the unmercifully small fields and the very few emergency fields. Everything connected with aviation in Madrasah at the present time is still very crude and weak handicapped by the ignorance of the owners of the engines and the mechanics. All that I have seen here on the grass land will have had just such people to deal with at some time or another."

Cross Country

Evergreen leaching soils are another great down-hill soil. There are a few that a pines could live on maybe safely, maybe not. They are mostly coarse, have and then with shrubs, herbs and grass, trees or a new growth of brush and vegetation. Should one add more soil, there is safety in one of these, probably, though there is no guarantee. It is not a soil that is safe to grow anything in. The coarse soil is washed out and will the few scattered native herb seeds probably be name of a hand-some as a horse as the people in general are very ignorant and forgetful. They think little of killing something, in the chance a Grizzly would have in a place like this, would very often do it. The few about up in the mountains follow the same pattern as the others. When the radius of forty rods I know of one very fair soil, though even a sterilized, and they are very moist and obstructed full of salts and high salts. We are near stone

San Pedro

The field at San Pedro's about forty miles from the coast, is only about six hundred feet in length and twice the width of the causeway, probably having a few acres of upland on the sides. It is, however, well taken care of, and has the best drainage, with a drain.

The other end and sides are surrounded by trees and the same old tropical growth. However, this field has the advantage of being only about a mile from town, and gas, oil, and natural gas are available.

There is a gas line running along the side of the field, but the gas is not being used now. Taking of there is a gas about forty feet deep to gas over right at the end of the field. There is a larger hot gas and one that is brought from town. There are no pipes in the field, and I have heard of some very bad gas burns, but I have not heard of any serious ones.

During the fall I made a trip by train to La Ceiba, about forty miles east of Tela, to locate a holding field there. This field in more or less prepared and should when completed be a fairly good place. There is another small place there that could be used if necessary. This field would naturally a bad place to store the planes and the holding fields are not large. I believe however the planes could be stored in the hangars at La Ceiba.

The next day we discussed again the question of the emergency port. I shall answer the latter first. We have had no trouble getting space as I have needed it and it takes about two or three months to get them when they are wanted. There were a few parts needed in the dock which were ordered Sept. 15 and Feb. 12 I received notice that they were not to be ordered at once. I have had enough trouble with the port authority to get them now. I have arranged through the concession to get that under approval as it will probably take another three months to get the parts. I believe this though, and when the new wings were bought there were also plenty of spares for all ordinary use.

Mechanics

The pilot must do precisely all the mechanical work, at least he must know how to do it. Then there is a mechanical time, a man whose experience is largely on gasoline motorcars, and on the gasoline tank. He calls the extremes "them" and "us". And all this is to be done in a short time, for the emergency is there, and the man is looking for an answer. Even to the filling of the gas or gas or water, every act must be watched all the time. The inevitable does happen. When the gasoline is taken and two or three tanks have been opened, perhaps to add from this field, the result of the time may be seen in Japan.

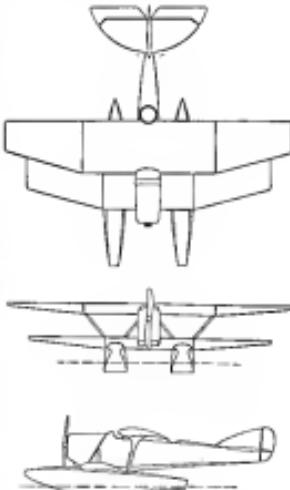
We live in Taile, a United Fruit Company port, and a good deal like a small mining town in the States. The Americans all live in a colony, houses all prettily alike, and at home there would be considered very inferior even for mint sheep. They also own the dairy, truck garden, cannery, in fact

July 27, 1965

AVIATION

everything, and anything that is needed must be bought from these at a very disadvantageous price. Living in a foreign country in a place that has no atmosphere of a small American town city is not nice. The enchantment of a foreign land is all lost, everything being perfectly flat and uninteresting.

The floats are 38 ft. long and weigh only 305 lb. apiece. They are of rectangular shape and are covered with three-ply. The shape is modified somewhat after that of the Brasdellberg floats and have a flat bottom forward and V bottom toward the rear.



Outline description of the Parchers Indians

being a political agent also. Should I seem a bit un-suited it is not because I want to be, but that my desire is to tell the truth about all these things. Flying at its worst in the States, is much better than flying at its best in Blondiana, any day.

New French Experimental Seaplane

The Presidore Co. has recently built for the French navy an experimental type of seaplane, which it is claimed is very simple to maintain and to assemble. The machine is in two parts, fitted with a 400-hp. Lorraine-Dietrich engine and is designed for observation and photographic work.

The engine is a four-cylinder, air-cooled, vertical type, with 566 cu. ft. of displacement, the lower wing resting directly on the frame, with no intermediate struts, and the upper wing resting directly on the top of the fuselage. The wings are of the "middle-wing" type with plywood covering. There are only a few struts in the upper wing, the main ones being two bracing struts from the fuselage to the upper wing. The fuselage extends from the front of the engine to the rear of the upper wing.

change of the power plant where they may be mounted on the pontoons while a single strut runs out from the pontoons to the upper wing. The upper wing with a span of 49 ft. 2 in. is quickly demountable so that the plane can be stored in ship board.

In response to many requests, we are able to announce that the Quiet Birdmen have not discontinued but will resume at once. *Nature and Science* will be

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